

1 The effects of wind-induced inclination on the dynamics of
2 semi-submersible floating wind turbines in the time domain

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16 **Abstract**

17 This study focusses on the coupling effects arising from the changes in the hydro-
18 dynamic behaviour of a semi-submersible floating wind turbine when it undergoes large
19 inclinations under wind loading. By means of a range of time-domain simulations, it is
20 shown that both the hull geometric nonlinearity effect and the alteration of viscous hy-
21 drodynamic forces can significantly affect the dynamics of a typical floating wind turbine
22 operating in waves at rated conditions. The consequences of said effects for both aligned
23 and misaligned wind and waves are explored. In general terms inclinations are found to
24 increase motions, where the modes that are more affected depend on the relative direction
25 between incident wind and waves. Understanding the sources of aero-hydrodynamic cou-
26 pling is key to providing sound design and modelling guidelines for the coming generation
27 of floating wind turbines.

28 *Keywords:* floating, wind, turbine, semi-submersible, large-angle, inclination

29 **1. Introduction**

30 In recent years floating wind power has been increasingly regarded as an attractive
31 option for the production of low-carbon electricity, thanks to the potential to unlock vast
32 resources which are unexploitable using fixed substructures; these are expected to become
33 gradually unviable for depths beyond 50-60 m [1, 2]. Being able to deploy wind turbines
34 in deep water will be crucial to determine the scale of the industry within regions where
35 the maritime continental shelf is steep. In spite of the presence of vast shallow areas

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36 especially in the North Sea, an estimate of the technical resource potential in Europe
37 indicates a deep-water share of about 70% [3]. Estimates for France range between 60%
38 [4] and 80% [3]. In Japan, now a prominent country in floating wind developments, 80%
39 of the offshore wind resources are located in deep water according to [5].
40 Different from most conventional offshore floating structures, floating wind turbines
41 (FWTs) are relatively small bodies which can exhibit stronger nonlinearities in their
42 dynamic behaviour. Moreover, they are designed with the purpose of maximising the
43 aerodynamic interaction related to wind energy extraction, which gives raise to unusually
44 large aerodynamic load to displacement ratios. This constitutes an important source of
45 dynamic coupling, especially as FWT platforms tend to evolve toward more optimised,
46 lightweight solutions. Characterising the mechanical behaviour of a floating wind turbine
47 for design and verification purposes requires the coupling of wind turbine aerodynam-
48 ics with offshore hydromechanics. The understanding of such coupled dynamics under
49 complex met-ocean loading has recently been the driver of a novel generation of coupled
50 offshore dynamic models designed for the requirements of FWT mechanical simulation,
51 such as FAST [6, 7, 8], HAWC2 [9, 8], FloVAWT [10], Simo-Riflex [11, 8], and CAL-
52 HYPSON of EDF R&D, the software used in the present study.

53 *1.1. Small offshore structure hydrodynamics*

54 Compact floating platforms can exhibit increased hydrodynamic complexity when
55 subjected to ocean waves compared to their larger counterparts; for example it is more
56 likely to come across regimes where hydrodynamic drag plays an important part in ex-
57 citation, as it was observed experimentally on the DeepCwind-OC4 platform by [12],
58 and explained numerically in [13]. These phenomena typically affect structures featur-
59 ing sharp-edged motion control devices, tanks, and pontoons, which accentuate flow
60 separation. Surface proximity effects can also manifest on these appendices when their
61 submergence is limited, such as increased vertical wave loading [conjectured in 14] and
62 run-up [15]. As shown by the experimental campaign carried out by [16] on a CALM
63 buoy equipped with a skirt, a semi-empirical numerical model implementing linear poten-
64 tial diffraction/radiation and a Re-independent drag force formulation can satisfactorily
65 (but not comprehensively, as explained in 1.2) represent the hydrodynamic forces acting
66 on this type of structure for the calculation of dynamic response. Similar conclusions
67 have been drawn by [17] whilst comparing numerical and experimental motion results for
68 a compact water-injection platform concept, the predecessor of the WindFloat platform
69 design. An analogous numerical-experimental comparison carried out for the engineer-
70 ing design of WindFloat itself broadly confirmed the accuracy of this type of numerical
71 model [2]. Next follows a brief close-up on water entrapment device hydrodynamics and
72 the main related modelling challenges.

73 *1.2. Water entrapment plates*

74 The water entrapment principle, often utilised in the hydrodynamic design of FWTs,
75 provides a passive motion control tool through the installation of relatively low-cost ap-
76 pendices. Pioneered by Principle Power with the WindFloat prototype, the heave plate
77 appendix consists in a thin reinforced structure installed coaxially below the platform's
78 columns, as visible in Figure 1. The dynamic stability provided by the use of heave
79 plates, coupled with the extra static stability insured by a closed-loop active ballasting



Figure 1: Detail of a WindFloat prototype column. Photo courtesy of Principle Power.

80 system, reportedly allowed the WindFloat prototype to adopt conventional aerogenera-
 81 tor technology [18].

82 The modelling of water entrainment appendices close to the free surface via linear dif-
 83 fraction and radiation plus a drag model should come with a caveat. As pointed out
 84 by [16], the radiation-dependent vertical added mass of these structures is suspected
 85 to suffer from the irrotational flow hypothesis (i.e. the model fails to take into account
 86 the momentum transfer needed to impel fluid rotation around the edges, causing under-
 87 estimation of added mass). Another issue consists in the sensitivity of the separation
 88 pattern to flow regimes, and in particular to KC [19]. The resulting drag forces — which
 89 dominate the hydrodynamic damping for this type of platform — may be affected by
 90 such regime changes, thus requiring appropriate adjustments of the drag coefficient. Fi-
 91 nally, nonlinearities caused by complex phenomena such as wave decomposition [20] and
 92 breaking [21] over the plates may perturb loading in ways that are not captured by the
 93 most widespread wave-structure interaction models.

94 1.3. Large inclinations

95 One of the routes to FWT CAPEX reduction is the compression of platform fabrica-
 96 tion cost. An immediate consequence of this is the push for the minimisation of platform
 97 mass and hence size, that in turn entails the availability of smaller displacements and
 98 waterplane areas for the sake of hydrostatic stability. Subsequently, low hydrostatic sta-
 99 bility platform solutions are currently being proposed. One option is constituted by TLPs
 100 [see for example 22], whose restoring capacity to oppose the aerodynamic overturning
 101 forces is built into the mooring system. An alternative approach is simply the accep-
 102 tance of large-angle operation caused by limited stability, leading to the introduction of
 103 the highly compliant FWT concept [23, 24]. This, combined with other technological
 104 considerations, has caused a range of tilt-tolerant floating VAWT designs to be spawned
 105 (see [25] for a technical discussion and [26] for an industrial application). Although con-
 106 ventional HAWT rotors are known to be tilt-adverse — especially with respect to their
 107 aerodynamic efficiency — angles up to 10° are beginning to be considered acceptable as
 108 the operational limit for this type of turbine [see 27, 28].

109 Several widespread assumptions of offshore structure dynamic simulation are challenged
 110 by the allowance of relatively large angular displacements. First of all, the ubiquitous
 111 hydrostatic linearisation may undermine the correct representation of these forces, espe-

112 cially when the geometry around the waterline is complex and/or hull sides are inclined
 113 (see for example the WINFLO concept [29]). The classic static representation of the
 114 mass matrix in the inertial frame can also cause errors in the computation of inertial
 115 reaction forces as angles break the small displacement assumption. Also, the classic
 116 linear superposition of small rotations may prove inaccurate, an observation that has
 117 led to the development of FWT motion solvers applying sequential Euler angle changes
 118 to represent correctly the nonlinear coupling between motions for a rigid-body [30, 31]
 119 and a multi-body system [32, 33]. Finally, the combination of limited draft, significant
 120 inclinations, and the presence of hydrodynamically sensitive appendices — typical of
 121 semi-submersible FWTs unequipped with active wind load compensation — has been
 122 shown to hold significant potential for the appearance of geometric nonlinearities in the
 123 diffraction/radiation behaviour of the hull [34]. The present work builds upon these find-
 124 ings, focussing on the effects of large inclinations on FWT dynamic response due to the
 125 alteration of both inertial and viscous hydrodynamic forces. Compared to the preceding
 126 work carried out on this subject, in this study a time-domain implementation enables an
 127 integral representation of drag forces as well as the inclusion of a coupled, yet simplified,
 128 aero-gyroscopic module representing the wind turbine rotor and tower forces.

129 2. Methodology

130 A program named CALHYPSO (*CALcul HYdrodynamique Pour les Structures Off-*
 131 *shore*) has been developed at EDF R&D that incorporates the aerodynamic, hydrostatic,
 132 hydrodynamic, inertial, and mooring forces exerted on a FWT to resolve its dynamics in
 133 the time domain. Whilst HAWT aerodynamics are here represented via a reactive thrust
 134 module, CALHYPSO also features a fully-fledged double-multiple streamtube represen-
 135 tation of VAWT aerodynamics. This simulation tool has been both verified and validated
 136 in the past; simple code-to-code verification examples are available in [35], whilst the ex-
 137 perimental validation of some of the inclination effects described here is scheduled for
 138 publication in 2016 (two related articles have been proposed to [36]). The combination
 139 of modelling methods adopted in this study, constituting a subset of the CALHYPSO
 140 framework, will be explained in the present section.

141 2.1. Coupled dynamics

142 The FWT dynamics are implemented using a 6-DoF, rigid body assumption with
 143 the reference point O placed at the centre of the structure’s waterplane area. Although
 144 the choice of this point is entirely arbitrary for the representation of rigid-body dynam-
 145 ics, when floating structures are concerned the above choice of O allows to adopt the
 146 widespread metacentric representation of hydrostatic stiffness without incurring in er-
 147 rors. Based on Newton’s second law of motion, the EoM of the system are written in the
 148 inertial system of reference (SoR) centred in O in the form:

$$\begin{aligned}
 (\mathbf{M} + \mathbf{A}^\infty)\ddot{\mathbf{x}}(t) &= \mathbf{f}^h(t) + \mathbf{f}^e(t) + \mathbf{f}^r(t) + \mathbf{f}^v(t) + \\
 &+ \mathbf{f}^a(t) + \mathbf{f}^g(t) + \mathbf{f}^m(t),
 \end{aligned}
 \tag{1}$$

149 where \mathbf{M} denotes the rigid body’s generalised inertia matrix, \mathbf{A}^∞ the infinite fre-
 150 quency added inertia matrix, and $\ddot{\mathbf{x}}$ the acceleration vector. The RHS term will be

151 explicitated in the next paragraphs. It consists in the summation of the instantaneous
 152 hydrostatic forces \mathbf{f}^h , the hydrodynamic excitation forces \mathbf{f}^e , the wave radiation forces
 153 \mathbf{f}^r , the viscous hydrodynamic forces \mathbf{f}^v , the aerodynamic forces \mathbf{f}^a , the rotor gyroscopic
 154 forces \mathbf{f}^g , and the mooring restoring forces \mathbf{f}^m . This equation allows to calculate $\ddot{\mathbf{x}}$ at
 155 each time step, then the time domain motion history is obtained by double numerical
 156 integration of the acceleration vector.

157 2.2. Hydrostatic forces

The hydrostatic force is computed using the conventional sea-keeping linearisation:

$$\mathbf{f}^h(t) = -\mathbf{K}^h \mathbf{x}(t). \quad (2)$$

158 By assigning a seawater density ρ and calculating the waterplane area A_w , the third
 159 diagonal hydrostatic stiffness coefficient is obtained as $K_{33}^h = \rho g A_w$. By denoting the
 160 displacement Δ and the transverse and longitudinal metacentric heights over the CoG
 161 respectively \overline{GM}_T and \overline{GM}_L , the roll and pitch hydrostatic stiffness coefficients are writ-
 162 ten as $K_{44}^h = \Delta \overline{GM}_T$ and $K_{55}^h = \Delta \overline{GM}_L$. The remaining entries of the \mathbf{K}^h matrix are
 163 nil for the type of structure studied.

164 2.3. Hydrodynamic forces

165 As it is frequently done in floating structure dynamic modelling, hydrodynamic forces
 166 are calculated via a superposition of inertial and viscous forces. The former are com-
 167 puted by NEMOH, an open source linear diffraction and radiation solver developed by
 168 the Ecole Centrale de Nantes [37], and imported within CALHYPSON in the form of a
 169 frequency-domain hydrodynamic base. This is used in turn to recreate, in the time do-
 170 main, the incident wave excitation vector \mathbf{f}^e and the reactive force associated to wave
 171 radiation \mathbf{f}^r in the form reported next.

172 The j^{th} element of the wave excitation vector is defined for an N -component, bi-dimensional
 173 wave train of incidence θ as

$$\begin{aligned} f_j^e(t) = \sum_{n=1}^N D_{j,n} a_n \cos(-\omega_n t + k_n \bar{x}(t) \cos \theta + \\ + k_n \bar{y}(t) \sin \theta + \Phi_{j,n} + \phi_n) \end{aligned} \quad (3)$$

174 where $D_{j,n}$ and $\Phi_{j,n}$ are the amplitude and phase of the n^{th} excitation harmonic
 175 in the j^{th} DoF, a_n the amplitude of the corresponding spectral component (or simply
 176 the incident wave amplitude if $N = 1$), ω_n and k_n the wave component's frequency and
 177 number, (\bar{x}, \bar{y}) the structure's mean horizontal offset, and ϕ_n a randomly generated phase
 178 ($\phi_n = 0$ if $N = 1$).

179 The radiation force vector \mathbf{f}^r is included using the linear impulsive model by [38] via the
 180 convolution integral

$$\mathbf{f}^r(t) = - \int_{t-T}^t \mathbf{K}(t-\tau) \dot{\mathbf{x}}(\tau) d\tau, \quad (4)$$

181 where T is the convolution window length and \mathbf{K} the convolution kernel, which is
 182 derived from the frequency-dependent radiation damping matrix [39].

183 Viscous forces are computed by discretising the hull into segments and evaluating the
184 quadratic drag force acting on each wetted element applying the Morison formulation
185 [40], proportionally to the projected area of these elements. The axial and transverse
186 components of the local relative velocity are derived from the incident wave kinematics
187 and the structure’s motion. The adoption of axial and transverse drag coefficients —
188 dependent on element geometry — allows to calculate the corresponding components of
189 the drag force. The transverse drag coefficients for all cylindrical sections are assigned
190 based on a (KC, β) pairing representative of an average regime, using the experimental
191 data provided by [41]. The axial drag coefficients for the heave plates are adopted from
192 [42], where such values are calibrated for thick water entrapment plates based on basin
193 tests. The elementary drag forces are integrated over the hull to form the viscous force
194 vector $\mathbf{f}^{\mathbf{v}}$ at every time step.

195 2.4. Aerodynamic forces

196 The aerodynamic forces acting on the rotor and the tower are both assimilated to
197 a thrust-type force, applied punctually at the rotor’s and tower’s respective centres of
198 thrust. The aerodynamic torque exerted on the rotor is also computed. Given an oper-
199 ating condition characterised by a thrust coefficient c_T and a torque coefficient c_Q , the
200 thrust and torque on the rotor are computed using a modified version of the coupled
201 formulation used by [43]:

$$T = \frac{1}{2} c_T \rho_a A |U - u| (U - u) \cos \alpha, \quad (5)$$

$$Q = \frac{1}{2} c_Q \rho_a A R |U - u| (U - u) \cos \alpha, \quad (6)$$

202 where ρ_a denotes the air density, A the rotor swept area, R the rotor radius, U the
203 incident wind speed at hub height, u the component of the hub velocity in line with the
204 incident wind speed, and α the instantaneous leeward inclination of the rotor due to plat-
205 form motions. This coefficient-based reactive model assumes that blade pitch does not
206 adjust to the aerodynamic fluctuations due to the motions of the FWT. This corresponds
207 to an idealised implementation of a known control option for floating HAWTs where the
208 frequency of the controller is deliberately moved below that of FWT motions, in order
209 to avoid the appearance of negative damping [44]. The thrust T is applied horizontally
210 in the direction of the incident wind and its moments are calculated with respect to
211 the EoM reference point using the updated position of the hub. Q is also applied as a
212 horizontal torque vector in the wind direction (it is assumed that the rotor is perfectly
213 yawed into the wind at all times).

214
215 The use of a coefficient-based model ensures that the rotor’s aerodynamic excitation
216 tensor matches a prescribed (equilibrium) operating point, based on the specifications
217 supplied by the turbine designer. However, apart from the averaging of rotor forces and
218 the absence of the effects of turbine control, an important limitation of such models is
219 the absence of a feedback mechanism relating the inflow speed to the fluctuations of T
220 and Q . This causes an overestimation of the aerodynamic forces’ variance in presence
221 of windward/leeward motion which is particularly significant at low frequency, when the
222 inflow has sufficient time to react. An example is given in Figure 2, where the thrust

223 component in the x direction for a rigid NREL 5 MW rotor oscillating in the direction of
 224 the incident wind is shown as predicted by FAST¹ and by the present simplified model.

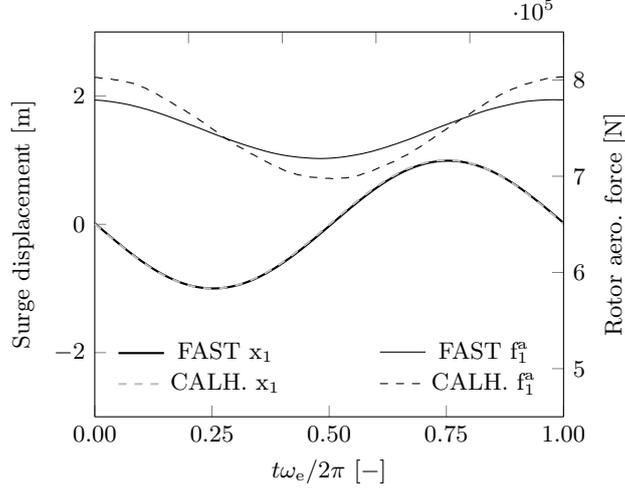


Figure 2: Forced oscillation in the direction of wind of an operating NREL 5 MW rotor, for an excitation frequency of $\omega_e = 0.40$ rad/s.

225 The incident wind speed and operating parameters of the turbine are set at the rated
 226 point and a constant wind profile is used. The Figure shows how the coefficient-based
 227 model provides the correct average thrust and a good approximation of the shape and
 228 phasing of its evolution over time, but produces a significantly larger variance than the
 229 more sophisticated blade element momentum theory.

230 Finally, the thrust exerted on the tower is also computed with Equation 5, using $c_T = c_D$,
 231 the drag coefficient for a cylindrical section. In this case A denotes the tower's projected
 232 area and u the component of the motion-induced velocity of the centre of thrust in the
 233 direction of the incident wind. The set of generalised aerodynamic forces resulting from
 234 the above model are summed and included in the term denoted \mathbf{f}^a .

235 2.5. Rotor gyroscopic forces

236 Past studies revealed the importance of gyroscopic coupling in floating wind dynamics
 237 [46, 47]. An oscillating structure bearing a rotor revolving at constant speed receives a
 238 gyroscopic reaction moment \mathbf{M}^g that can be written using the d'Alembert principle [48],

$$\mathbf{M}^g(t) = -I_r \Omega \mathbf{\Gamma}(t) \times \mathbf{i}^r(t). \quad (7)$$

239 In the above equation I_r represents the axial inertia of the rotor, Ω its rotational
 240 speed and \mathbf{i}^r the associated unit vector, whilst $\mathbf{\Gamma} = (\dot{x}_4, \dot{x}_5, \dot{x}_6)$ denotes the structure's
 241 rotational speed vector. The gyroscopic term included in Equation 1 is then written as

$$\mathbf{f}^g = (0, 0, 0, M_1^g, M_2^g, M_3^g). \quad (8)$$

¹FAST v.8.08, using dynamic inflow. The rotor speed and blade pitch are fixed. Surge motion is obtained with the application of a harmonic horizontal force as proposed in [45].

242 *2.6. Mooring forces*

243 CALHYPSO includes the capability for a multi-segment, quasi-static representation
 244 of catenary moorings and their forces [a similar model is described in 49]. In the present
 245 study mooring restoring forces are represented with a linearised model employing the
 246 stiffness matrix \mathbf{K}^m ,

$$\mathbf{f}^m(t) = -\mathbf{K}^m \mathbf{x}(t), \quad (9)$$

247 in order not to incur in offset-dependent mooring stiffness nonlinearity and response
 248 bifurcation.

249 *2.7. Treatment of mean inclination*

250 The approach adopted here to treat the FWT's relatively large wind-induced inclina-
 251 tions is based on re-linearising the dynamic system about the tilted and offset configura-
 252 tion assumed by the FWT under pure wind loading. It is assumed that further dynamic
 253 oscillations of small amplitude will occur around this position.

254 A preliminary computation is required to obtain said offset configuration: in this study
 255 this consists in applying the rotor forces obtained with the desired operational regime
 256 and running the dynamic simulation in the absence of incident waves until the steady-
 257 state, static offset is attained. Small static rotations are found about the z axis (around
 258 0.2°) and about the horizontal axis aligned with the wind (around 0.4°). These are
 259 respectively due to the eccentricity of the thrust force with respect to the z axis and
 260 the stator's reaction to the rotor torque. These secondary rotational components are
 261 neglected whilst the larger leeward equilibrium angle due to wind overturning is used to
 262 rewrite the EoM terms with the methods explained next.

263 As it will be pointed out in Section 4, because of the nonlinearities present in the model
 264 the effective mean inclinations in the wind-and-wave dynamic simulations do slightly
 265 depart from the static values obtained as described above. Although this problem may
 266 be solved iteratively, in the present study this relatively small discrepancy between input
 267 and output mean inclination is accepted.

268 *Rigid body inertia.* The generalised inertia matrix is actualised to the mean rotated
 269 position using

$$\mathbf{M} = \begin{pmatrix} m & 0 & 0 & 0 & mz_G & -my_G \\ m & 0 & -mz_G & 0 & 0 & mx_G \\ & m & my_G & -mx_G & 0 & 0 \\ & & J_{11} & I_{12} - mx_G y_G & I_{13} - mx_G z_G & \\ \text{sym.} & & & J_{22} & I_{23} - my_G z_G & \\ & & & & & J_{33} \end{pmatrix}, \quad (10)$$

$$\begin{aligned} J_{11} &= I_{11} + m(y_G^2 + z_G^2), \\ J_{22} &= I_{22} + m(x_G^2 + z_G^2), \\ J_{33} &= I_{33} + m(x_G^2 + y_G^2), \end{aligned} \quad (11)$$

270 where the terms denoted I are the elements of the central inertia tensor. Vector
 271 (x_G, y_G, z_G) represents here the new position of the centre of gravity in the inertial SoR
 272 after applying the rotation.

273 *Hydrostatics.* A set of preliminary calculations determined that the changes in hydro-
274 static stiffness never exceed 1% for the treated hull and its mean inclinations. For vertical
275 hydrostatic stiffness they consist in computing the updated waterplane area, whilst for
276 the rotational terms the positions of G (centre of gravity), B (centre of hull volume), and
277 M (longitudinal/transversal metacentre) are recalculated after applying an isocarenic in-
278 clination. Thus their heights over the reference keel point K concur to determining the
279 updated hydrostatic restoring moment arm $\overline{GM} = \overline{KB} + \overline{BM} - \overline{KG}$ and finally the stiffness
280 terms K_{44}^h and K_{55}^h as described in 2.2. Following these considerations, it was decided
281 to neglect the nonlinear hydrostatic effects due to the mean inclination, (which is in the
282 order of 6° in the presented case study). For what concerns the rotational terms, the
283 above finding is consistent with the well-known behaviour of wall-sided floating bodies,
284 which exhibit linear hydrostatic restoring up to trim/heel angles of about 10° [see for
285 instance 50]. Thanks to the choice of point O at the SWL, no extra-diagonal terms
286 appear within the \mathbf{K}^h matrix following a static inclination.

287 *Potential flow hydrodynamics.* The geometric nonlinearity affecting the hull's wave dif-
288 fraction and radiation in presence of a significant inclination can be treated by updating
289 the hydrodynamic mesh to the new mean position, effectively re-linearising inertial hydro-
290 dynamic forces about a new equilibrium point. Figure 3 shows the upright hydrodynamic
291 mesh and the actualised meshes following application of wind overturning in the x and y
292 direction. Slender cross-beams are omitted from the potential flow hydrodynamic mesh
293 for computational efficiency reasons. It can be noticed that in the presence of a rotation
294 about the x axis the xz planar symmetry is lost, leading to the need to discretise the
295 entire wetted surface. An important remark based on this figure is that the water entrap-
296 ment plates undergo significant vertical excursion in (b) and (c), causing the potential
297 flow hydrodynamic regime perturbations already scrutinised in [34]. Their impact on
298 FWT dynamic response in the time domain will be treated in Section 4.

299 *Viscous hydrodynamics.* The process of calculating the hydrodynamic drag forces in-
300 cludes updating the position of the discrete hull elements at every time step as a conse-
301 quence of the motion of the structure. Thus the preliminary computation of the mean
302 inclination described above needs not be an input, since the correct drag element dis-
303 placements are continuously applied in the time domain. This implies that the drag
304 elements are exposed to wave particle kinematics of varying intensity depending on their
305 vertical excursion and of varying phasing depending on their horizontal excursion. Fig-
306 ure 4 provides a visual example of the lateral drag force exerted on a platform column
307 undergoing excursion.

308 *Mooring forces.* All nonlinearities related to the catenary mooring system are neglected
309 to keep the focus on platform hydrodynamics, hence the initial linearised stiffness matrix
310 \mathbf{K}^m is employed unchanged.

311 3. Definition of case study

312 To keep the focus sharp, this study adopts the modified Dutch Tri-floater, a FWT
313 concept elaborated by [51] and modified by [34], which is particularly suited to show the

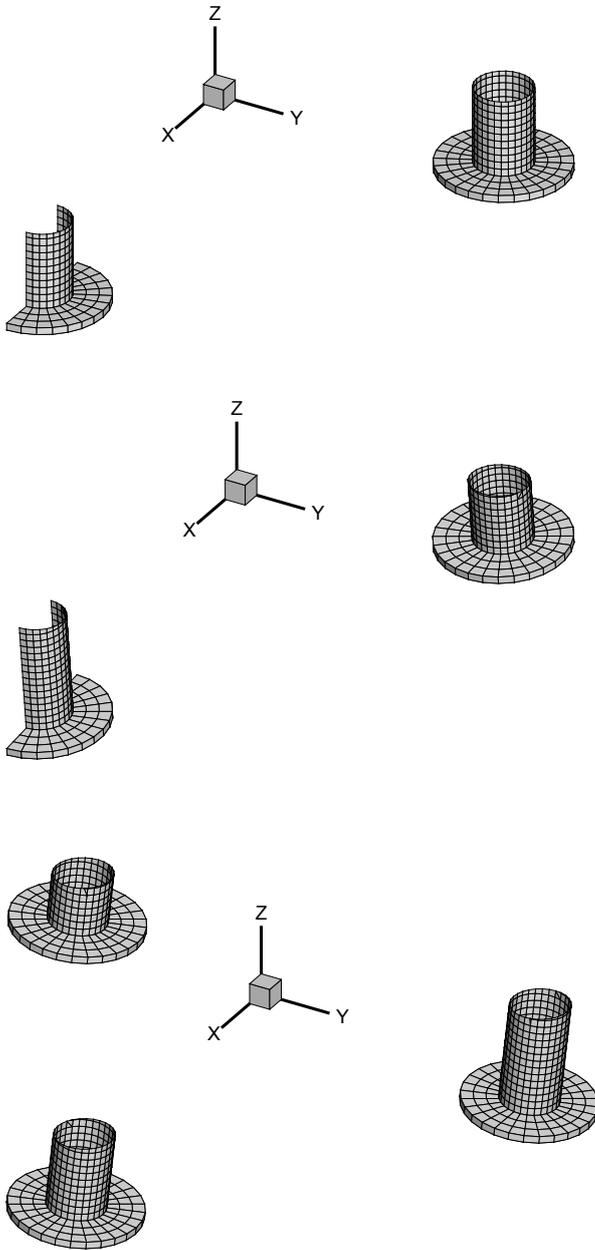


Figure 3: Wetted surface discretisations used for the wave diffraction and radiation calculations. Above: upright. Centre: with rotation about y . Below: with rotation about x .

314 effects of wind-induced inclination on coupled dynamic response. This structure is sub-
 315 jected to inline and cross wind and wave regimes using a homogeneous wind profile and

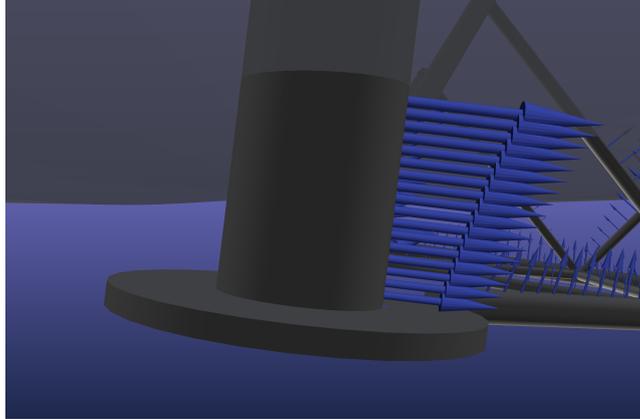


Figure 4: Snapshot of transverse drag forces exerted on a platform column under a 12 s wave of 4 m height. The arrow size is proportional to the magnitude of the local force.

316 regular or irregular waves as inputs. In the irregular wave cases, the standard JONSWAP
 317 formulation is used to calculate the input spectrum, employing $\gamma = 3.3$. All wavetrains
 318 propagate in the positive x direction, exciting the structure in presence of zero, inline,
 319 and cross wind at rated speed, constant over time. The wind turbine's operating condi-
 320 tions are parked for zero wind speed and rated for the cases including wind. A compact
 321 definition of the loading cases considered is supplied in Table 1.
 322

Table 1: Definition of loading cases. Angles are measured in the horizontal plane counter-clockwise starting from the x axis (90° is aligned with y).

Regular waves			
Loading case name	LC0	LCX	LCY
Wind speed [m/s]	0.0	11.4	11.4
Wind direction [deg]	-	0.0	90.0
Turbine operation	parked	rated	rated
Wave height [m]	4.0	4.0	4.0
Wave direction [deg]	0.0	0.0	0.0
Irregular waves			
Loading case name	LC0i	LCXi	LCYi
Wind speed [m/s]	0.0	11.4	11.4
Wind direction [deg]	-	0.0	90.0
Turbine operation	parked	rated	rated
Sig. wave height [m]	4.0	4.0	4.0
Peak period [s]	10.0	10.0	10.0
Wave direction [deg]	0.0	0.0	0.0

323 The three-column semi-submersible floating platform is coupled to a NREL 5 MW
 324 reference offshore wind turbine [52], totalling 3124 t of displacement (Figure 5).

325 Station keeping is assured by a conventional chain-wire hybrid, 6-point mooring sys-

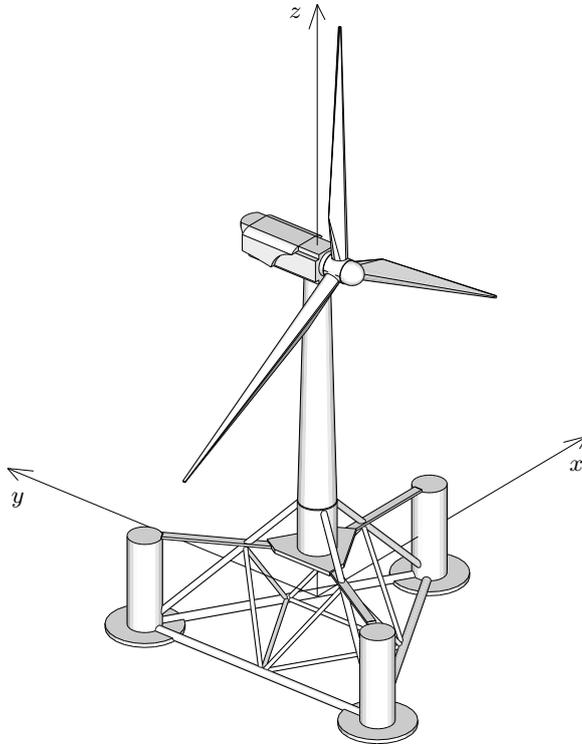


Figure 5: Modified Dutch Tri-floater with 5 MW class horizontal axis wind turbine. The latter is adapted from a graphical model made available by [53].

326 tem (line properties are given in Table 2) with two lines departing from the outer bottom
 327 of each column, at a radial distance of 4.0 m from the centre of the column. A water
 328 depth of 50 m is assumed.

Table 2: Properties of a Dutch Tri-floater mooring line.

Component	studless chain	wire
Material	steel	steel
Nominal diameter [m]	0.15	0.16
Unstretched length [m]	190	225

329 The parameters defining the platform are given in Table 3, whilst the turbine is
 330 detailed in Table 4.

331 As said the mooring restoring forces are included with a simple restoring term; the
 332 undisturbed mooring weight, equating 183.5 t, is accounted as a lump mass at fairlead
 333 height within the weight computation. Equation 12 defines the values assigned to the
 334 mooring stiffness tensor, as calculated by [51]. The units used are [m, rad] for displace-
 335 ments and [N, Nm] for the generalised forces.

Table 3: Modified Dutch Tri-floater parameters. Mass and inertia quantities are expressed with respect to the origin O at SWL.

Geometry	
Design draft [m]	12.0
Hull volume at design draft [m ³]	3048
Column centre-to-centre spacing [m]	68.0
Column diameter [m]	8.0
Column depth incl. plate [m]	24.0
Plate diameter [m]	18.0
Plate thickness [m]	1.0
Bracing diameter [m]	1.0 to 2.0
Mass & inertia	
Mass [t]	2263
Vertical position of CoG [m]	-0.1
Roll/pitch moment of inertia [t m ²]	$1.535 \cdot 10^6$
Yaw moment of inertia [t m ²]	$2.522 \cdot 10^6$

Table 4: Adapted 3-blade upwind 5 MW NREL offshore wind turbine data. Mechanical quantities are expressed with respect to the origin O at SWL with the exception of the rotor's inertia.

Mechanical parameters	
Overall mass [t]	678
Overall vertical position of CoG [m]	83.0
Overall roll/pitch moment of inertia [t m ²]	$3.779 \cdot 10^6$
Overall yaw moment of inertia [t m ²]	$5.220 \cdot 10^3$
Rotor axial inertia [t m ²]	$3.876 \cdot 10^4$
Elevation of tower/platform interface [m]	25.0
Rotor diameter [m]	126
Hub height [m]	90.0
Rated operating parameters	
Incident wind speed [m/s]	11.4
Rotor speed [rpm]	12.1
Rotor thrust coefficient c_T [-]	0.82
Rotor torque coefficient c_Q [-]	0.066
Tower drag coefficient c_D [-]	1.0

$$\mathbf{K}^m = \begin{pmatrix} 1.6 \cdot 10^5 & 0 & 0 & 0 & 1.9 \cdot 10^6 & 0 \\ & 1.6 \cdot 10^5 & 0 & -1.9 \cdot 10^6 & 0 & 0 \\ & & 1.5 \cdot 10^5 & 0 & 0 & 0 \\ & & & 1.1 \cdot 10^8 & 0 & 0 \\ & \text{sym.} & & & 1.1 \cdot 10^8 & 0 \\ & & & & & 1.7 \cdot 10^8 \end{pmatrix} \quad (12)$$

336 **4. Results**

337 This section presents the results of the set of simulations defined above. The prelim-
 338 inary offset calculation yields the generalised position vectors

$$\begin{aligned} \text{LCX \& LCXi: } \mathbf{x} &= (3.96 \ 0.08 \ 0.00 \ 0.41 \ 6.37 \ 0.18)^T \\ \text{LCY \& LCYi: } \mathbf{x} &= (-0.08 \ 3.96 \ 0.00 \ -6.37 \ 0.41 \ 0.18)^T \end{aligned} \quad (13)$$

339 with displacements given in [m] and rotations in [deg]. As one may expect, the most
 340 significant static offsets are a displacement in the wind direction, which is resisted by
 341 the mooring system, and a leeward rotation mainly countered by hydrostatic restoring.
 342 The symmetric stiffness features of the system also emerge from Equations 13. The
 343 FWT's dynamic response in regular waves exhibits nearly harmonic form at steady state.
 344 Obtaining the pseudo-transfer function of motion in any DoF i requires post-treating the
 345 time-domain signal, which in this case is done by

$$r_i(\omega) = \frac{\max[x_i(t, \omega)] - \min[x_i(t, \omega)]}{H}, \quad (14)$$

346 where $H = 2a$ is the incident wave height. It must be pointed out that because of the
 347 nonlinearities present in the model, the pseudo-transfer function magnitude is dependent
 348 on the excitation magnitude within certain bands. In case wind forces are applied, the
 349 response characteristics r_i are to be interpreted as the normalised amplitude of the dy-
 350 namic response to wave excitation about the mean wind-induced offset.

351 Figure 6 displays the ensemble of the pseudo-transfer functions calculated for the loading
 352 cases defined in Table 1. For each DoF the corresponding uncoupled, undamped natural
 353 frequency ω_n is reported to facilitate interpretation. It can be seen that the most impor-
 354 tant response features lie in the xz plane (surge, heave, pitch), which is expected for a
 355 2D wavetrain propagating in the x direction over a roughly symmetric structure. Most
 356 resonances are confined at the far left of the studied band, a desirable hydromechanic
 357 feature in light of the distribution of wave energy and the subsequent first order excita-
 358 tion. An in-depth screening of these results is given in Section 5.

359 Figure 8 provides a snapshot of a dynamic simulation in regular waves after the steady
 360 state is reached. The quasi-harmonic motion time histories of surge and pitch are shown
 361 with their respective aerodynamic excitation signals, for an incident wave frequency of
 362 0.40 rad/s and two different loading cases. The mean forces and by consequence the off-
 363 sets obtained when wind is applied are immediately evident. This case will be later used
 364 for clarification of the inter-DoF aerodynamic coupling observed around said frequency.

365 The motion outputs of the irregular wave simulations are shown in Figure 7 in spectral
 366 form. The underpinning time series have a duration of 2 hours and represent the fully
 367 developed dynamic response of the structure in a stationary sea state. The input spec-
 368 tral realisation — defined by a particular choice of component phases — is the same for
 369 all loading cases. These results will be used in Section 5 to contextualise the dynamic
 370 response features of the system for a specified, realistic met-ocean condition: the fact
 371 that most of the input wave energy lies between 0.5 and 0.9 rad/s brings out the response
 372 features of the studied FWT in this central band.

373 Finally, the mean leeward inclination angles attained in the steady-state phase of all dy-
 374 namic simulations are plotted (in absolute value) in Figure 9. It appears that although
 375 limited in magnitude, discrepancies exist between the assumed mean inclination and that

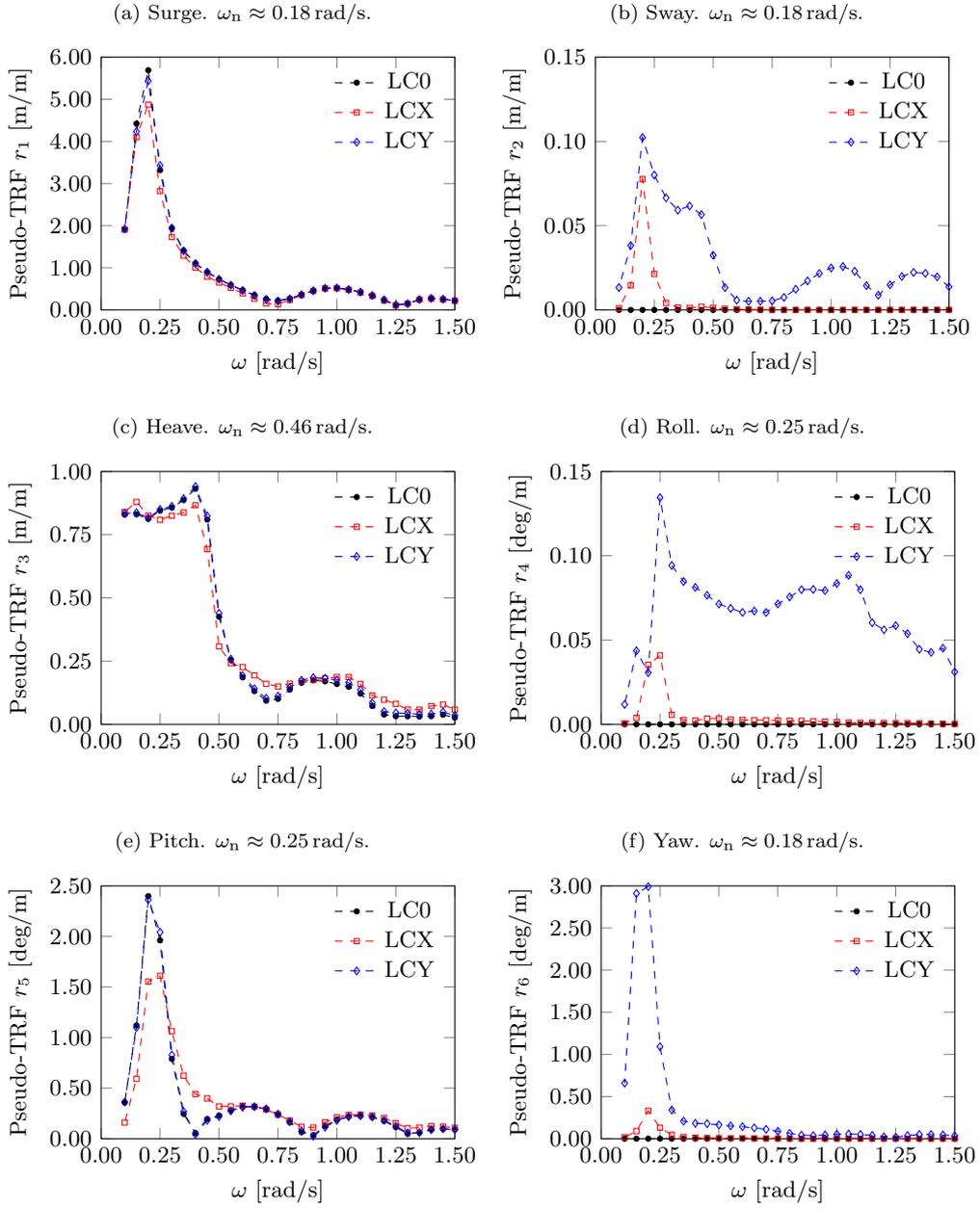


Figure 6: Pseudo-transfer functions of a modified Dutch Tri-floater subjected to wind and regular wave excitation.

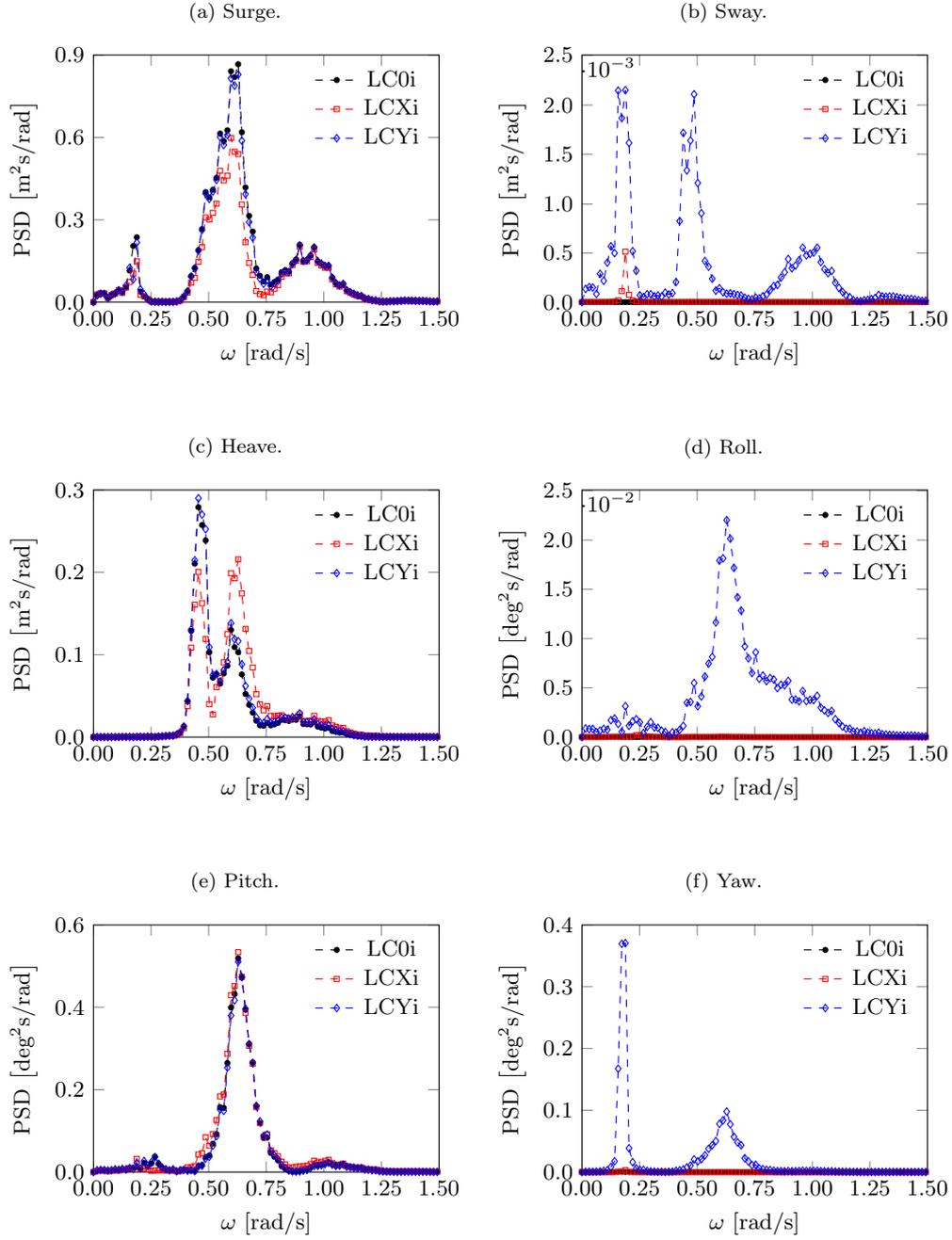


Figure 7: Power spectral density of the motions of a modified Dutch Tri-floater subjected to wind and irregular wave excitation.

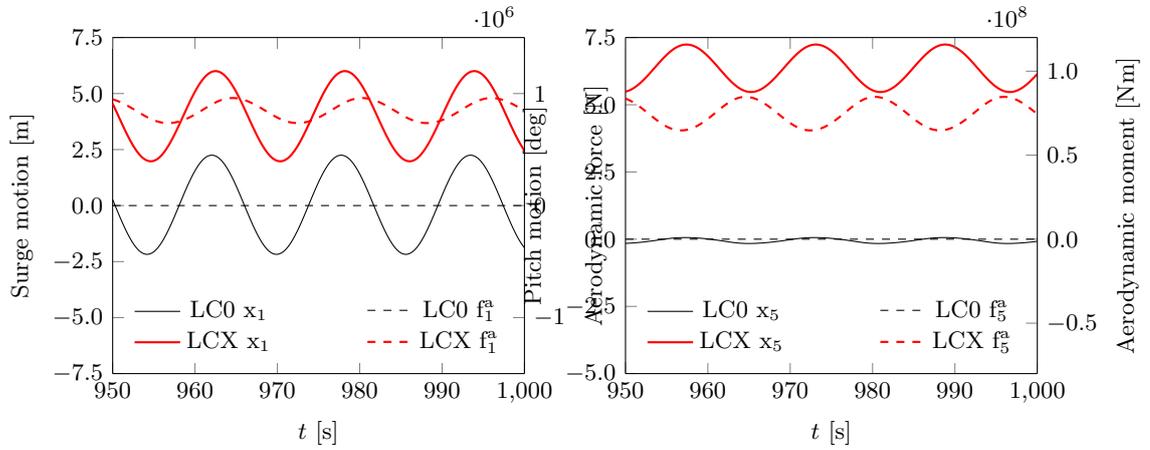


Figure 8: Time-domain output motion and aerodynamic forcing in surge (left) and pitch (right), for an incident wave of 0.40 rad/s.

376 effectively produced by the dynamic simulations in regular waves. This is especially pro-
 377 nounced for load case LCX, where nonlinear forces with non-zero mean are present (i.e.
 378 hydrodynamic drag) which have a large component in the same plane of the leeward
 379 inclination. The subsequent deviations from the statically calculated inclination do not
 380 exceed 4% and are deemed tolerable.

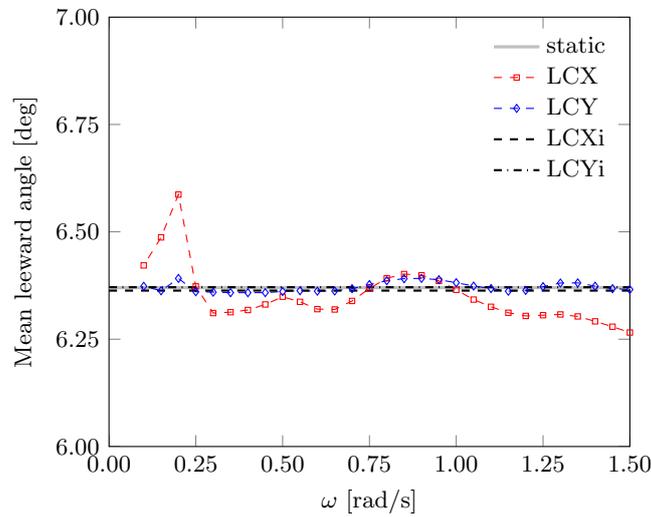


Figure 9: Mean leeward angles obtained from wind-and-wave dynamic simulations.

381 **5. Discussion**

382 Numerous elements of interest can be derived from the results shown in Section 4.
383 They will be discussed in the following by looking at a DoF at a time, then the key
384 findings will be related to existing experimental results.

385 *5.1. Dynamic response analysis*

386 Response in **surge** (Figure 6a) is not particularly affected by the presence of rotor
387 forces and platform inclinations. Only when the rotor is aligned with the surge motion
388 (LCX) a minor dampening effect occurs due to aerodynamic damping thereby reducing
389 the amplitude of the oscillation at the peak, located at 0.18 rad/s. The reduction in
390 off-peak response is instead related to the alteration of inertial hydrodynamic excitation
391 of the surge-pitch coupled mode. Observing Figure 7a reveals the significance of these
392 features in irregular waves.

393 As expected, and visible from Figures 6b and 7b, **sway** response is identically zero when
394 the structure receives wave excitation only (directed along the x axis). Resonant inter-
395 DoF coupling and loss of hydrodynamic xOz symmetry respectively cause small sway
396 response for collinear and cross-wind loading cases.

397 **Heave** motion (Figure 6c) exhibits a marked dependence on rotor forces and inclinations
398 in the case of collinear wind and waves: excitation suppression around $\omega \approx 0.7$ rad/s
399 becomes less pronounced mainly because of the simultaneous vertical excursion of two
400 heave plates toward the free surface (see Figure 3, centre). This in turn triggers

401 (1) an increase (and a phase shift) of the inertial wave excitation, an effect already
402 discussed by [34], and secondly

403 (2) an increase in hydrodynamic drag under the action of stronger wave kinematics.

404 Whilst effect (2) significantly contributes to altering response in said central band, effect
405 (1) is solely at the root of the increased response observed at higher frequencies. The
406 slight reduction of motion amplitude observable around the resonance peak is also an
407 effect of the large inclination due to heave-pitch coupling deriving from imbalance in wave
408 radiation forces (in the model: significant extra-diagonal terms appear in the added mass
409 tensor). The prevalence of these effects in the central band makes them significant in the
410 studied irregular wave scenario, as observable from Figure 7c.

411 **Roll** (Figure 6d) is another DoF whose response is nil in the upright position and in
412 the absence of rotor forces. Whilst the results are practically unchanged for LCX apart
413 from a small resonance effect, a broad, albeit limited motion response is present in cross-
414 wind conditions because of the loss of symmetry in port-starboard column excitation (see
415 Figure 3, below), which is also detectable in Figure 7d.

416 Figure 6e reveals that **pitch** response at low frequency is particularly affected by the
417 action of wind forces and the subsequent inclination in the collinear case. As expected,
418 the resonance peak around $\omega_n = 0.25$ rad/s is attenuated by the aerodynamic damping,
419 whose impact is otherwise insignificant in the cross-wind case. The increased response
420 manifested by LCX at the far right of the peak is exclusively due to the plate excursion
421 effect (1) described above. A most prominent feature of pitch motion for the collinear
422 loading case is the appearance of significant response in the immediate vicinity of the
423 resonance peak, where wave excitation is suppressed for LC0 and LCY. This is caused by

424 two concurring factors: heave plate excursion effect (1) and the aerodynamic excitation of
425 pitch caused by surge motion. This aerodynamically sourced, inter-DoF coupling will be
426 further clarified in 5.2. In the studied irregular wave case, these pitch response features
427 lose significance due to the low energy available below 0.5 rad/s (Figure 7e).
428 Figures 6f and 7f display the response in **yaw**, which rests unexcited by the waves in
429 the absence of wind. The combination of gyroscopic coupling and resonance produces a
430 limited response peak for LCX. It is the cross wind-and-wave cases LCY and LCYi that
431 display the largest motion: the immersion of the port column and the emersion of the
432 starboard column (Figure 3, below) cause an imbalance creating the potential for broad
433 yaw forcing across the studied frequency band. Where this excites resonant motion, in
434 the band around 0.18 rad/s, the dynamic response becomes very significant. It should
435 be remarked that since the present model omits the damping contributions of mooring
436 line drag and rotor yawing, peak yaw response is likely overestimated.

437 5.2. Aerodynamic inter-DoF coupling

438 For the presently studied FWT, the frequency band between 0.25 rad/s and 0.55 rad/s
439 is rich in inter-DoF coupling phenomena, which intertwine with the effects of wind-
440 induced inclination in a complex fashion. As pointed out in 5.1, aerodynamically sourced
441 surge-pitch coupling explains in part the observed difference in response between LCX
442 and the other regular wave loading cases (cf. Figure 6e). LCX is the sole case where
443 the rotor is operating in line with the hub speeds induced by pitch motion. Reactive
444 aerodynamic damping is a well-known consequence of this set-up. Within the frequency
445 band centred on 0.4 rad/s, however, wave-induced pitch motion is relatively small and the
446 aerodynamic force oscillations due to the motion-induced hub velocity tend to be more
447 associated with surge response. This is made evident by rearranging the LCX results of
448 Figure 8 (left) in terms of surge speed and aerodynamic force, as displayed in Figure 10
(left). Hence it emerges the relationship between surge and the aerodynamic force in the x

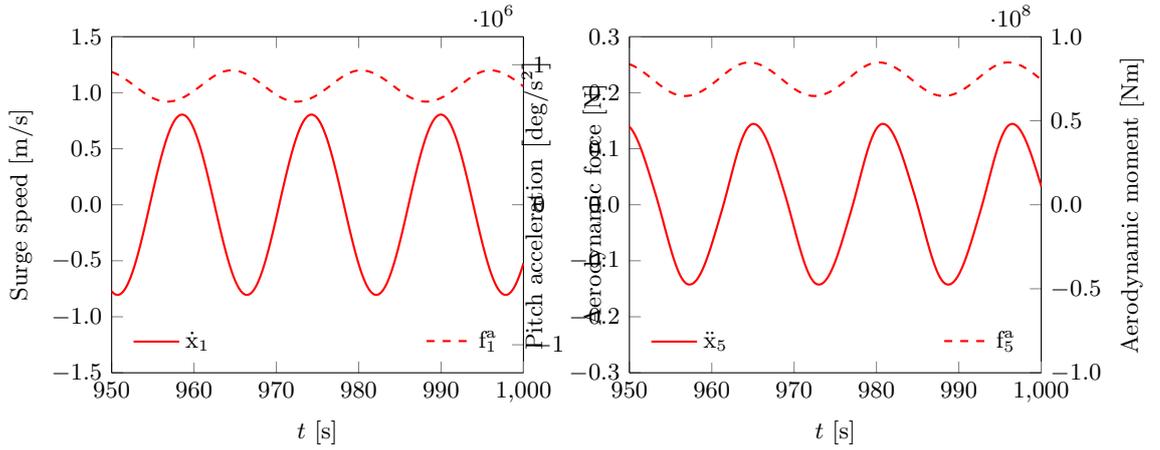


Figure 10: Time-domain surge speed (left) and pitch acceleration (right) of loading case LCX, for an incident wave of 0.40 rad/s. The respective aerodynamic force and moment signals are also shown.

450 direction: such force appears to react to surge motion since it is in near phase opposition
451 with surge velocity. At the same time, its fluctuations are transferred as variations
452 of external pitch moment to the FWT assembly. The phasing between aerodynamic
453 excitation and pitch acceleration visible in Figure 10 (right) for case LCX reflects this
454 inter-DoF coupling effect, as the aerodynamic reactions are driving, not dampening, pitch
455 motion. Of course, the retroactions present in the dynamic system close the loop and
456 ultimately render the one-way cause-and-effect dynamic explained above less clear-cut.
457 The effect of the rotor control strategy — conventional, low-frequency [44], or with active
458 motion damping [54] — on the intensity of this coupling is likely significant, although
459 not treated here. It should also be pointed out that since the present thrust-based model
460 tends to overestimate the aerodynamic reactive force (see Figure 2), the strength of this
461 coupling mechanism is likely overestimated as well, at least in the absence of active
462 motion damping by control.

463 *5.3. Experimental evidence of heave plate excursion effect*

464 The observed interaction between heave plates and wave forces in presence of large
465 vertical excursions is only accounted for by nonlinear fluid dynamic models or, to a
466 certain extent, by re-linearising a linear model about the offset position as done here,
467 thereby limiting the errors caused by geometric nonlinearity. A past publication by
468 [14] shows that the experimentally observed dynamics of a similar FWT subjected to
469 collinear wind and waves are not entirely captured using the conventional modelling
470 method based on linearising the fluid-structure interaction about the initial equilibrium.
471 Virtually equivalent results are also available in [51, 55]. The predictions of pitch — a
472 key DoF for FWTs — shown by these authors are particularly inaccurate in the 0.25 to
473 0.50 rad/s band, that is around the excitation suppression point next to the resonance
474 peak: said numerical models underestimate the relatively large pitch motion obtained
475 experimentally. On the contrary, the modelling strategy presented in this study may
476 enable a more accurate representation of FWT dynamics in this range, as suggested by
477 the sustained pitch response of LCX to the right of the main peak in Figure 6e. This
478 can also be shown by applying the conventional linearisation of potential hydrodynamics
479 to the above case study, which provides for instance the pitch response characteristic
480 provided with Figure 11. An ongoing simulation benchmark campaign is expected to
481 verify this capability by application of the present methodology to an experimentally
482 documented case.

483 **6. Conclusions**

484 The present work builds on previous findings which suggest that low-draft water en-
485 trapment plates undergoing excursion may alter the dynamics of a FWT when both
486 wind and wave loads are present [34]. The use of a time-domain model with inclination-
487 dependent, yet linear implementation of potential flow hydrodynamics enables to confirm
488 these features, detected among a range of aero-hydrodynamic coupling effects. Table 5
489 provides a summary of the observed couplings which descend from the mean wind-induced
490 inclinations (about 6°) attained when the turbine is operating at rated conditions. Both
491 inline and cross wind-and-wave cases are treated, showing that the geometric nonlin-
492 earity descending from the leeward inclination of the FWT significantly affects dynamic

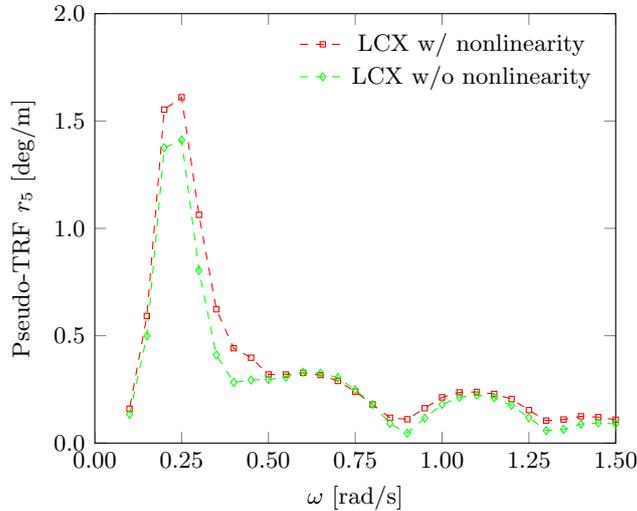


Figure 11: Pitch response characteristic predicted with and without treatment of the geometric nonlinearity.

493 response through the vertical excursion of the columns and the water entrapment plates.
 494 With respect to the direction of wave propagation, in-plane response (surge, heave, pitch)
 495 is mostly affected by inclination for collinear wind and waves, whilst out-of-plane response
 496 (sway, roll, yaw) is shown to be altered by the application of cross-wave wind.

Further work will be required to broaden the characterisation of the inclination effects

Table 5: Summary of the observed wind-induced inclination effects on FWT dynamics.

Inclination effect	Nature of force	Location	Affected DoF	Bandwidth	LC type
plate excursion	hydro. inertial	heave plates	heave, pitch	wide	collinear
	hydro. viscous	heave plates	heave	narrow	collinear
	hydro. inertial	heave plates	roll	wide	cross
column excursion	hydro. inertial	columns	yaw	wide	cross

497 to cover more operating conditions, different highly compliant FWT concepts (e.g. a soft
 498 spar), and possibly the interaction of these phenomena with wind turbulence and rotor
 499 control. A high-level classification of FWT concepts may also be performed, for instance
 500 by evaluating their motion sensitivity to the inclination effects as a function of governing
 501 parameters such as the wind load to displacement ratio.
 502

503 Given that the cost optimisation of FWT designs may keep pushing the allowable wind-
 504 induced inclinations further, the nonlinearities inherent to the coupled dynamic system
 505 are likely to increasingly manifest, more than anything else in the highly dynamic, di-
 506 mensioning loading cases. The dynamic modeller must then apply careful judgement:
 507 methodologies of increasing complexity will be required, likely departing from conven-
 508 tional offshore structure analysis. Although the present method can represent a compu-
 509 tionally efficient alternative to treating geometric nonlinearity, higher-order resolution

510 of fluid-structure interaction is likely required to accurately compute the hydrodynamic
511 loads in presence of larger inclinations. Nonlinear angular resolution of the EoM also be-
512 comes appropriate where angles exceed magnitudes of 10-15° and the linear superposition
513 of rotations is no longer accurate.

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